



CanSat Leader Training Program (CLTP) - 8<sup>th</sup> Cycle



# **Final Mission Report**

15/Sep./2017

Ryuichi Sekita

Department of Smart System School of Engineering Fukuyama University







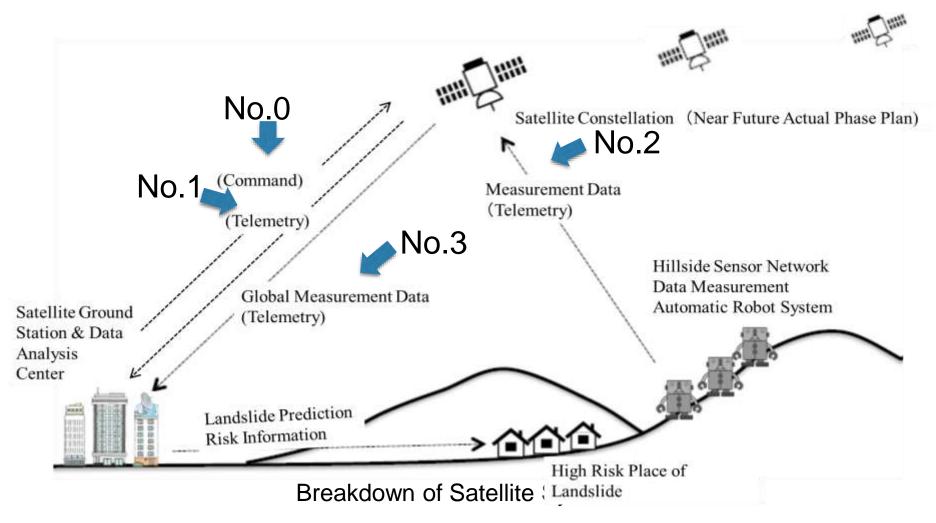


### Content

- Mission Statement
- Mission Requirements Validation and Verification Plan
- The Satellite System
- Bus System
- Payload Subsystem
- Mission Sequence (Result)
- Verification Sheet
- Flight Results First Attempt
- Flight Results Second Attempt
- Conclusion
- Recommendation and Future Plan (Mission)
- Feedback and Recommendation (CLTP)



### Mission Statement Extract Some Essence from My Research Mission



Do not Reproduce without Permission. Copyright: 2017 Nihon University



### **Mission Statement**

Mission No.0: Mission Ready Phase

Satellite should be Power Save Mode and Install Container Before drop, Set Ready to Do Mission by Full Power Command from GS

Mission No.1: Telemetry Gathering Phase Satellite should Gather Telemetry Data

(Acceleration, GPS data, Bat Vol. Temp.)

Mission No.2: Data Store Phase

Satellite should Receive the Random Number as Dummy Data from Ground (GS)

Mission No.3: Data Forward Phase

Satellite should Sending Picture Data and Telemetry Data to GS



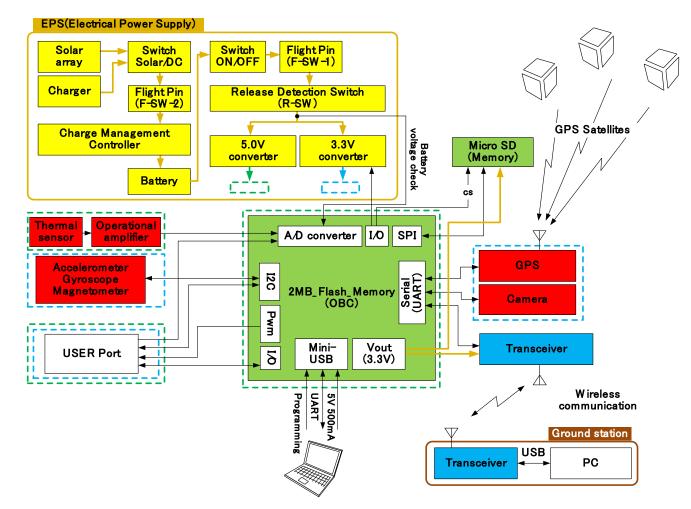
# Mission Requirements Validation and Verification Plan

No.	Event	Requirement	Function	Verification
1	Standby Phase	Battery Voltage is 4.0V or more	Battery Charge Function	Visual Inspection
2	Launch Phase	Set Power Save Mode Automatically by SW Set Full Power Mode by Command and Confirm Full Power in GS	Mbed Sequence GS Command Print Vout in GS	Soft Test Soft Test
3	Mission Phase	Receive Random No. Receive Snapshot Command Receive GPS signal Measure Accel. Temp. Volt. Endure Landing Shock	GS Command and Transceiver Function GS Command and Snapshot Function GPS Receiver Function and Telemetry Function Good Structure and Good Assembly	System TestSoft Test and Camera TestSystem Test and ATVisual Inspection
4	Analysis Phase	Final Presentation should be Good		Final Presentation Output



# Bus System Architecture

• No Additional Parts with Original HEPTA-Sat





# Payload Subsystem Architecture

- Original HEPTA-Sat Camera and Sensors
- No Additional Sensor and Components
- Software for both Sat. and GS should be modified to Appropriate Mission Function

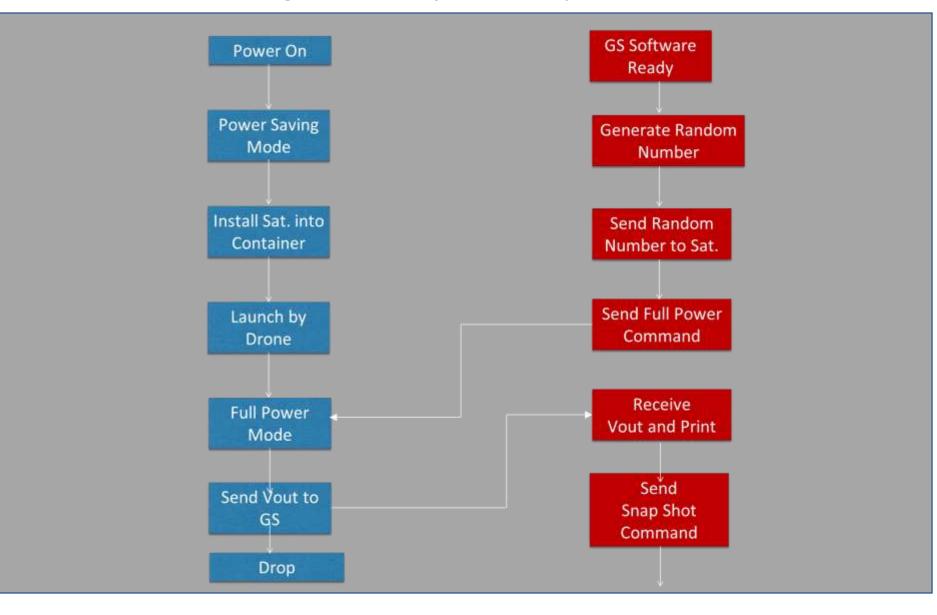


# **Verification Sheet**

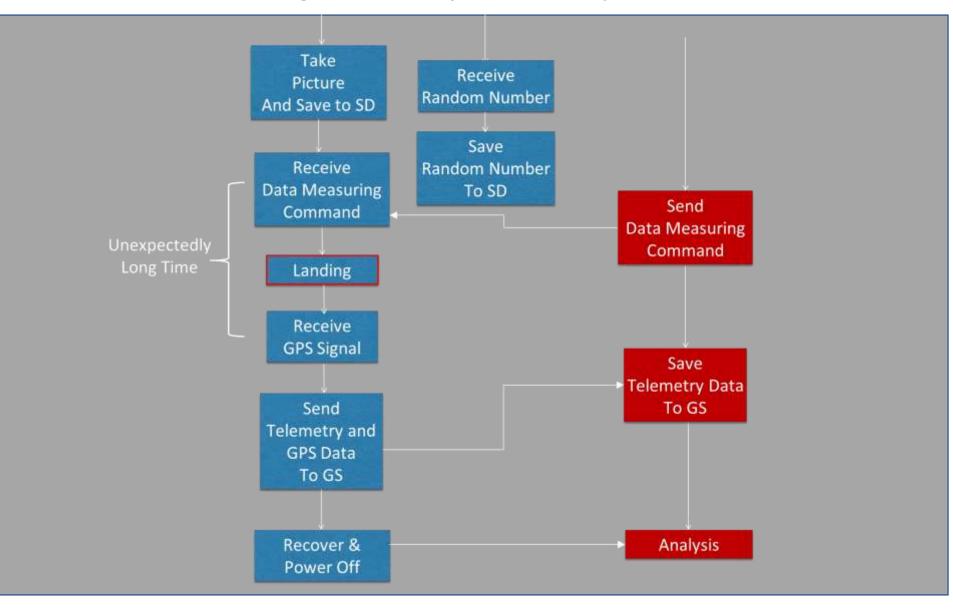
No.	Event	Requirement	Function	Verification	Result
1	Standby Phase	Battery Voltage is 4.0V or more	Battery Charge Function	Visual Inspection	3.5V (Not Charged) Replace New One
2	Launch Phase	Set Power Save Mode Automatically by SW Set Full Power Mode by Command and Confirm Full Power in GS	Mbed Sequence GS Command Print Vout in GS	Soft Test Soft Test	Good Good
3	Mission Phase	Receive Random No. Receive Snapshot Command Receive GPS signal Measure Accel. Temp. Volt. Endure Landing Shock	GS Command and Transceiver Function GS Command and Snapshot Function GPS Receiver Function and Telemetry Function Good Structure and Good Assembly	System Test Soft Test and Camera Test System Test and AT Visual Inspection	Partially Good (Not Solved) Good AT analysis is not finished yet Good, but 1 leg was Broken during Preparation Replace New One
4	Analysis Phase	Final Presentation should be Good	Sekita's Analysis Ability	Final Presentation Output	Good

### **Mission Sequence (Result)**





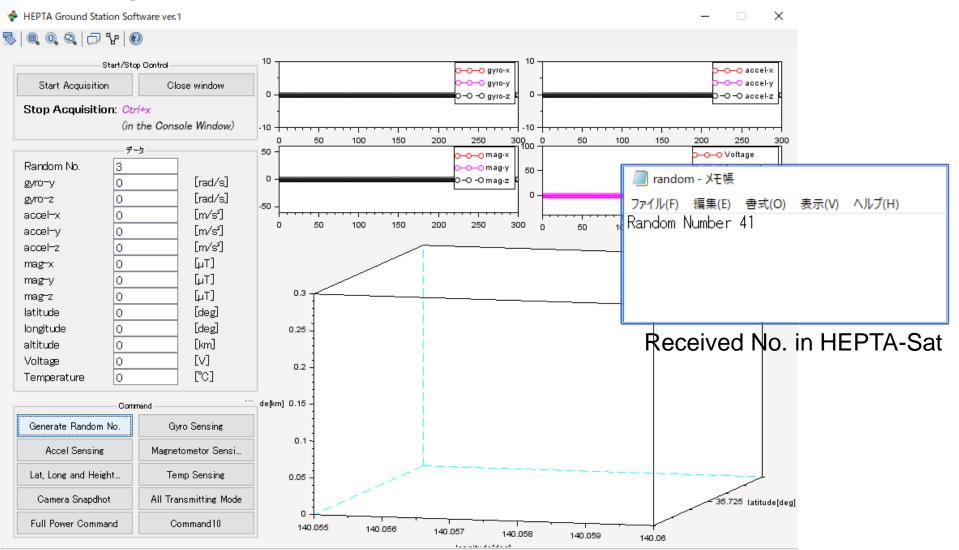
### Mission Sequence (Results)



Cansat Leader



### Flight Result: First Attempt





### Flight Result: First Attempt



#### **HEPTA-Sat Snap Shot**

# Flight Result: First Attempt

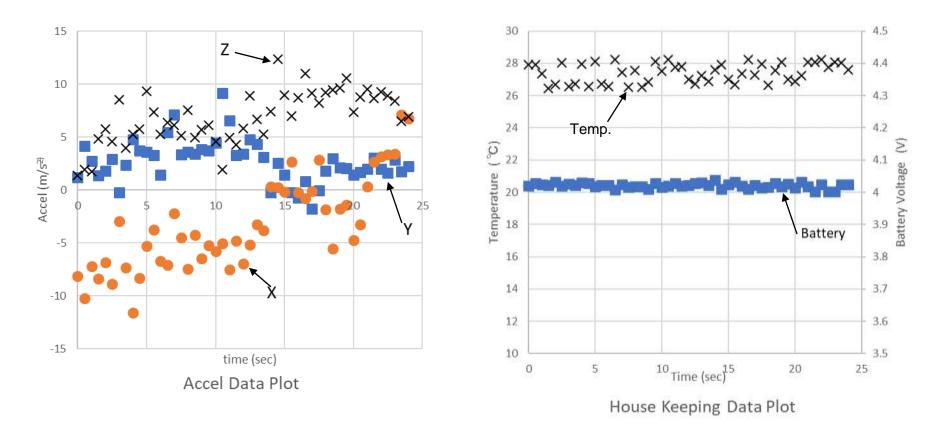




Do not Reproduce without Permission. Copyright: 2017 Nihon University



### Flight Result: First Attempt

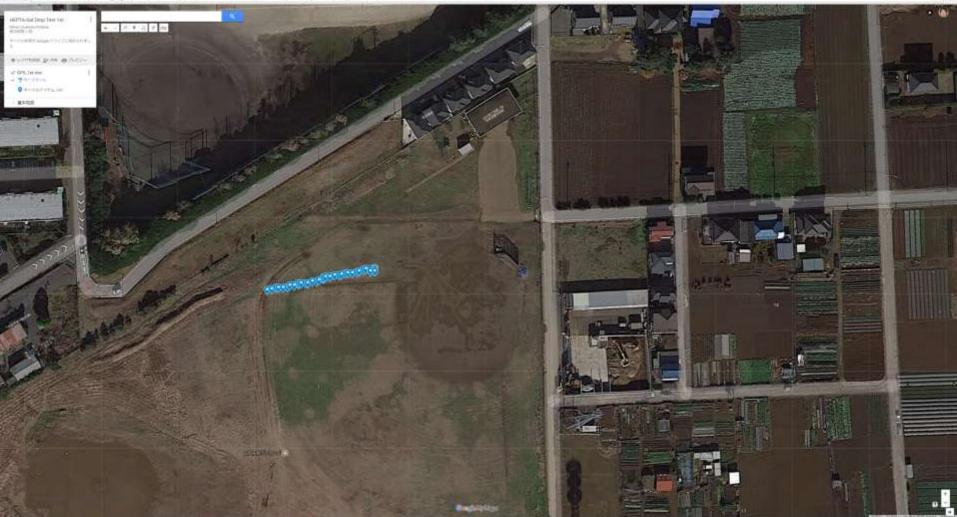




### Flight Result: First Attempt

annya V.T. II II III HAMA-Set Doop from the III II.

C + FRICE ( Interview goods on the Contraction of the Contract Con





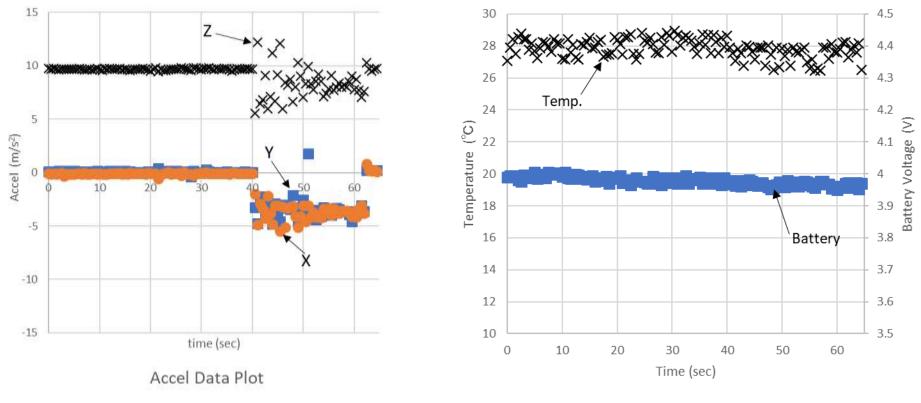
### Flight Result: Second Attempt



**HEPTA-Sat Snap Shot** 



### Flight Result: Second Attempt



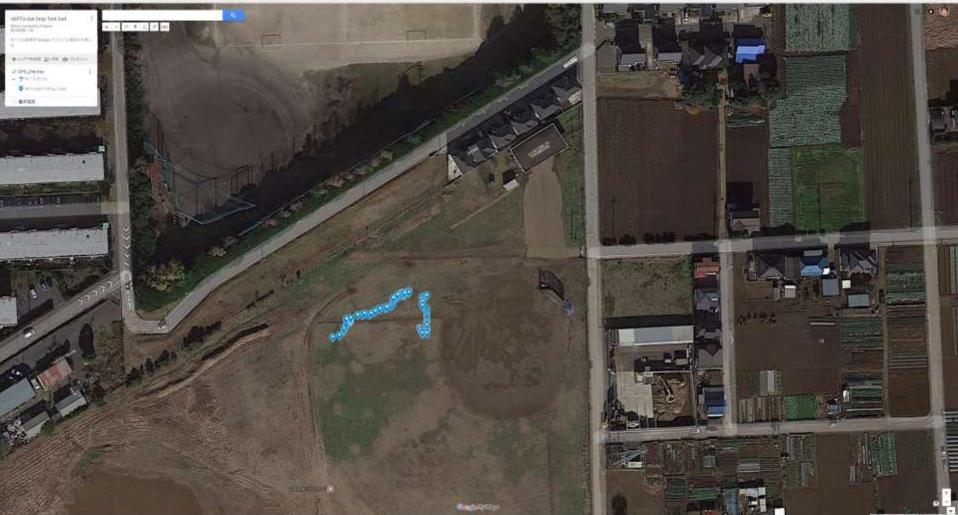
House Keeping Data Plot



### Flight Result: Second Attempt

Carrier Martin Party and Carrier Barrier and Carrier Barrier B

C ... Q 1 # R 2015 1 4 ( 100 / 200 /





# Conclusions

• My HEPTA-Sat System Achieve Minimum Mission Sucess

Success Level	Mission No.	Contents	Result
Minimum	Ready Phase 0	Satellite should be Power Save Mode and Install Container Set Ready to Do Mission by Full Power Command from GS Satellite should Gather Telemetry Data (Acceleration, GPS data, Bat Vol. Temp.)	Complete Complete Complete
Full	Data Stote 2 Data Forward 3	Satellite should Receive the Random Number as Dummy Data from GS Satellite should Sending Picture Data and Telemetry Data to GS	Complete Incomplete during Mission Time, but Complete after Landing
Extra	Upload GPS Track Data to Google Earth	Satellite should Receive GPS Raw data and Save in SD card	Incomplete using Raw Data, but Complete GPGGA Data to Google Map



# Recommendation and Future Work (Mission)

- AT Varidation
  - Calculate and Analysis of AT Data (Hexadecimal → Decimal → Physical Quantity)
  - Write Adequate Test Data Acquisition Code
- Quality Control and System Test Plan
  - Prepare for some System Trouble and Parts Failure
  - Develop Systematic Testing Plan
- Software Improvement
  - My MBED code must be improved to match GS Sending Random Number and HEPTA-Sat Receiving Random Number
- S&F Mission Simulation
  - Develop some Ground Sensor Data Receiving Function
  - Develop Full Automatic MBED Sequence after Receiving Full Power Command
- Build Up PBL Program using HEPTA-Sat



# Feedback and Recommendation (CLTP)

- Well Work out Program and Basically No Problem
- Two Option Models
  - No.1: Original Model (Easy to Assemble)
  - No.2: Soldering Necessary Model (Like a i-CanSat, Difficult to Assemble)
- There are some explanations that are difficult to understand and difficult to practice
- Printed Text Book is better than the Pad Version for me
- Is it enough to understand Systems Engineering and Management?
  - Probably, some R&D experience should be needed for the understanding



# Thanks for All ! & See You Again Near Future, R&D Site