

CanSat Leader Training Program (CLTP) - 8th Cycle

Final Presentation

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Mission Statement

Estimating Impact-Force of a Re-entry Vehicle For Hazardous Analysis and Structure improving



Mission success

- 9 axis sensor running and saving measurements
 to SD+ Temp +volt. <u>Minimum Success</u>
- 9 axis sensor running and transmitting measurements to ground station+Temp +volt

Success mission

Measuring impact force <u>Extra-Success</u>

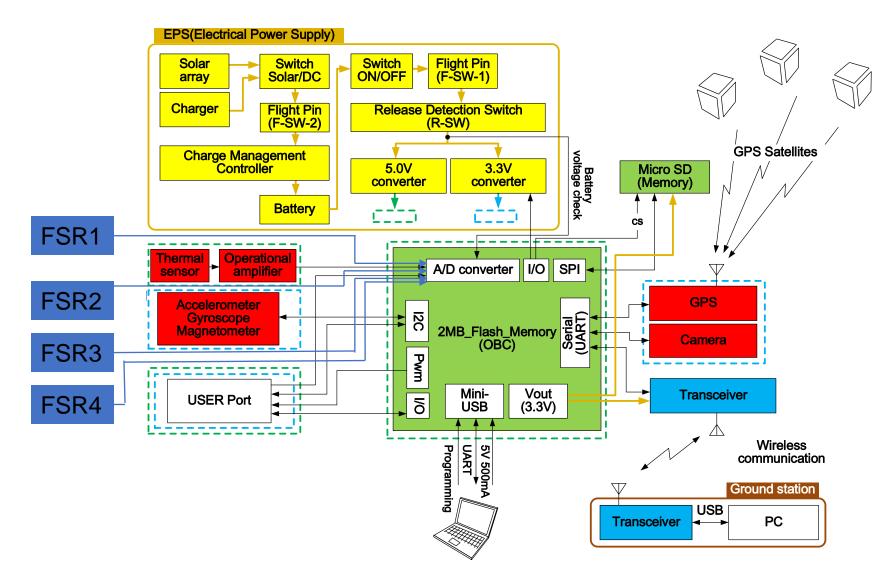
Mission Requirements



No.	Event	Requirement	Required Function	Verification Way
R-1	Preparation Phase	All systems having power	Check leds	All lights on All wires are in place
R-2		Wire are in place	Check wires	All whes are in place
R-3	Standby time phase	Battery Voltage is >4 volt	To charge battery from the external source	Confirm battery is charged using Multi- meter
R-4	Launch Phase	Install satellite on octocopter	Installation	Visual inspection
R-5			Run the octocopter	
R-6	Mission Phase	Release from octocopter	-	Visual inspection
R-7		Open parachute	Release command	
R-8		Getting the required data	Receiving data at Ground station	Data shows on Ground station
R-9		Landing	Manual landing of octocopter	Visual inspection
R-10	Analysis Phase	Structure checking and hazardous analysis	Structure analysis for structure and for human head	Results are consistent with scientific findings

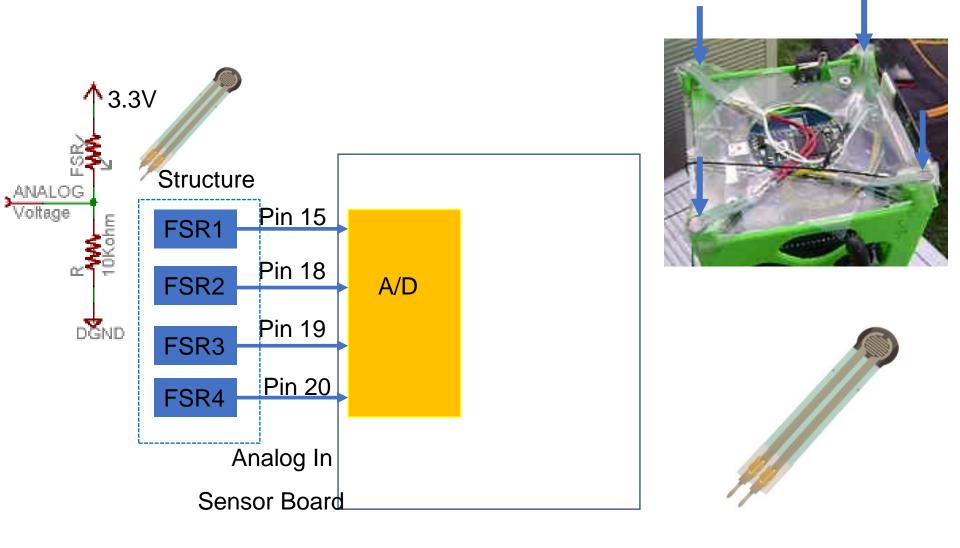


Satellite System Architecture



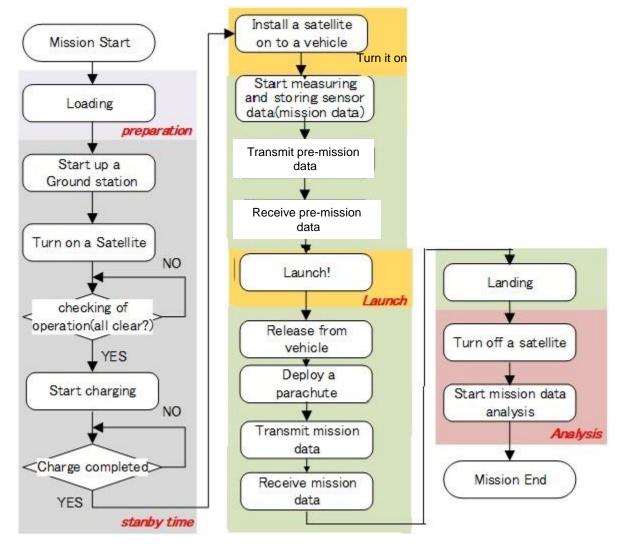


Payload Subsystem Architecture





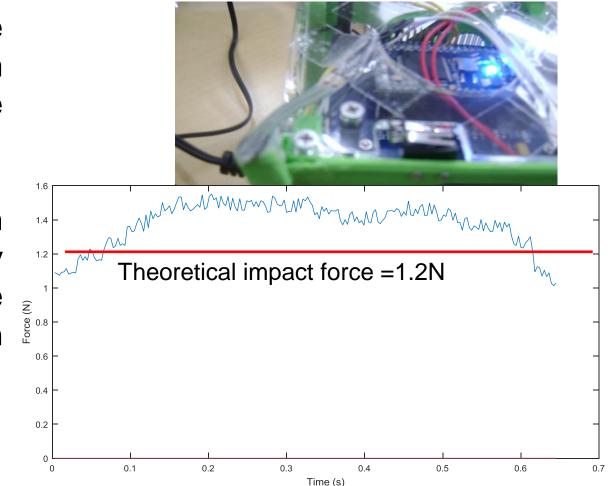
Mission Sequence

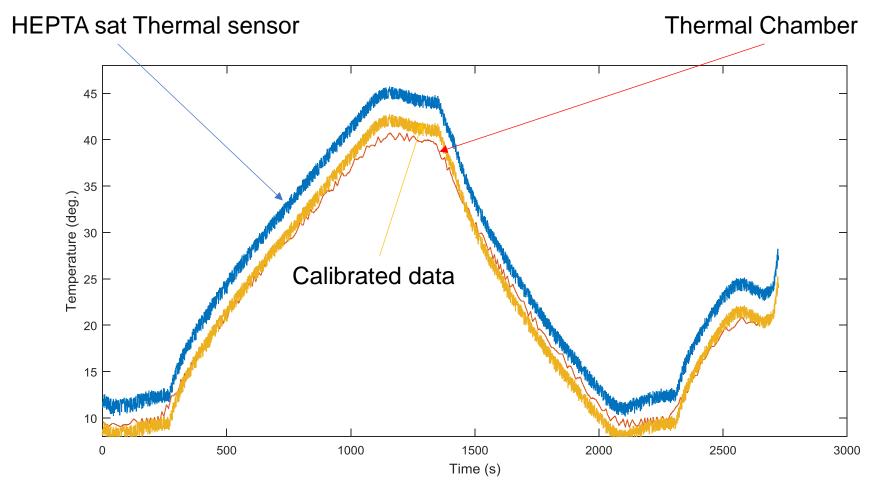


(1) Payload

The satellite were dropped from a 50 cm height and data were recorded.

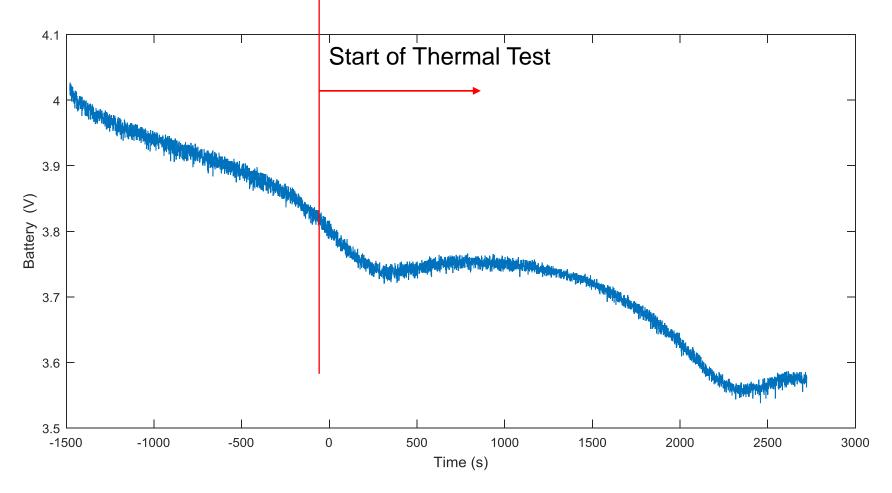
The load shows on one leg only and by inspection of satellite this leg was broken due to higher stress.



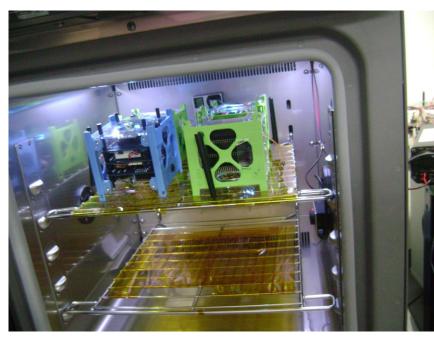


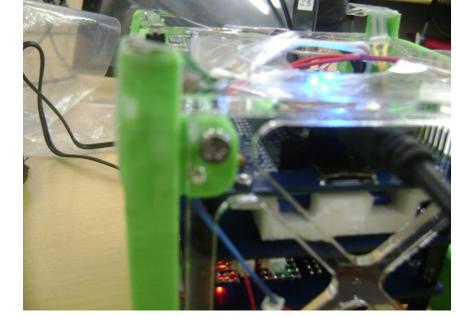
(2) Thermal sensor calibration

(3) Battery Endurance Check



(4) Communications with Xbee and data storage on the CD





Thermal Test

Drop Test



Flight Result: First Attempt

- •9 axis sensor running and saving measurements to SD. Minimum Success $\sqrt{}$
- 9 axis sensor running and transmitting measurements to ground station (seems slower than expected). Success mission $\sqrt{}$

• No forces were recorded for the four FSR sensors. <u>Extra-Success mission !! need to wait for analysis</u>



Flight Result: Second Attempt

- •9 axis sensor running and saving measurements to SD. Minimum Success $\sqrt{}$
- 9 axis sensor running and transmitting measurements to ground station (seems slower than expected suspecting interference). Success mission $\sqrt{}$

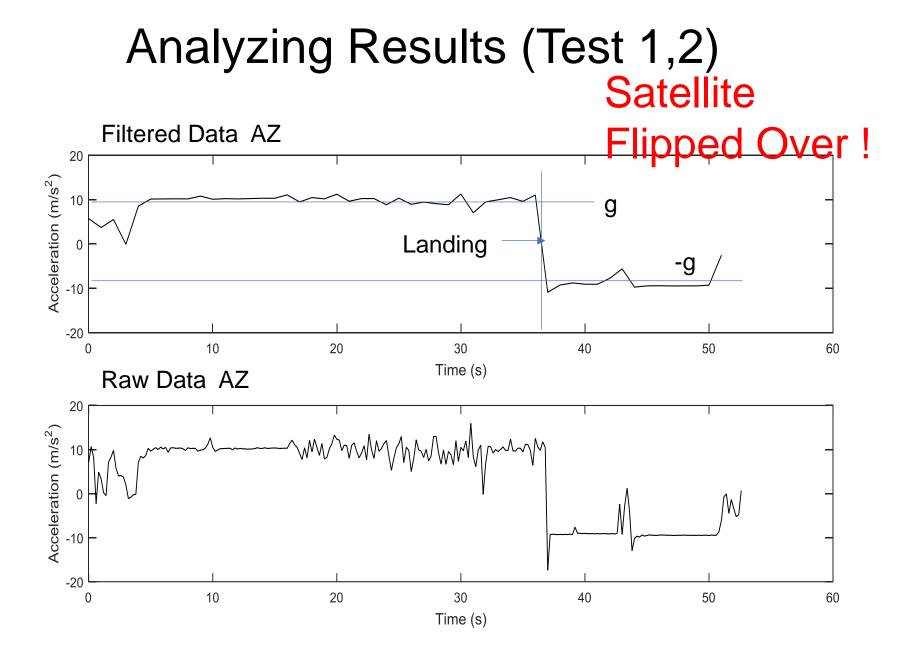
No forces were recorded for the four FSR sensors.
 <u>Extra-Success mission !! (need to wait for analysis</u>



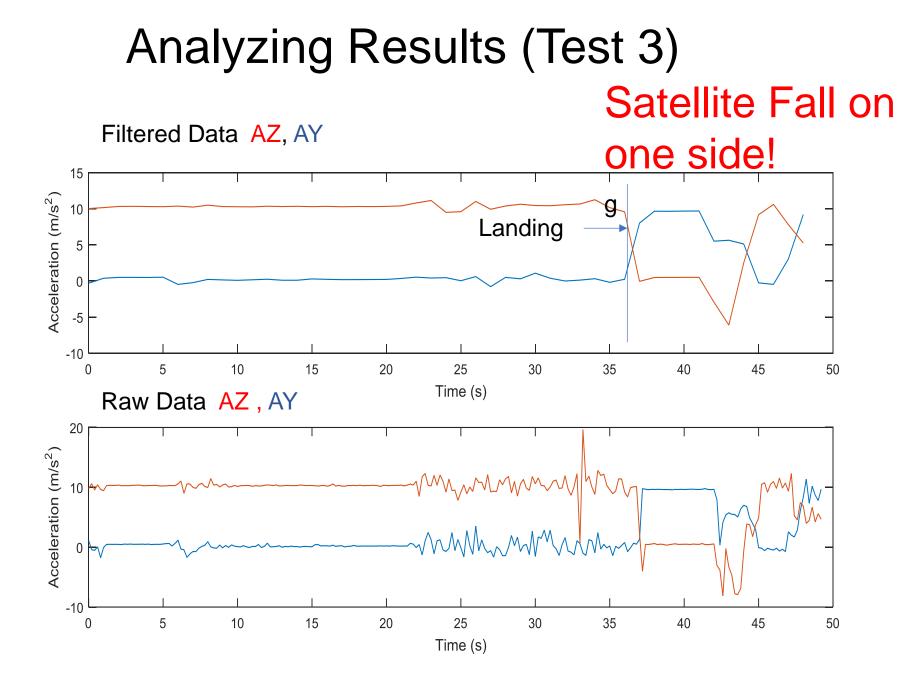
Flight Result: Third Attempt

- Keeping away from octocopter remote control.
- •9 axis sensor running and saving measurements to SD. Minimum Success $\sqrt{}$
- 9 axis sensor running and transmitting measurements to ground station. Success mission $\sqrt{}$
- No forces were recorded for the four FSR sensors.

Extra-Success mission !! need to wait for analysis



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Controlled drop on Grass

I did several drops on the grass from random altitude and could get one reading each drop ranging from 400gf to 1500gf (4-15N)





Conclusions

- •FSR are working fine for calculating impact forces under controlled experiment (drop test).
- In real flight nothing can guaranty that the satellite will land on its
- •legs.
- Impact force can break
- the leg of the satellite.





Recommendation and Future Work (Mission)

• For FSR to work in general flight test, we need

more on all sides of the satellites.

- I would like to model the satellite and the parachute motion dynamics.
- I would like to try different experiments and sensors



Feedback and Recommendation (CLTP)

- Structure of the Hepta need improvement
 - (strengthening the legs and parachute connections.
- A day or two more for the training period will be appreciated.

